TO: ALL HOLDERS OF HORIZONTAL STABILIZER CENTER ELEVATOR CONTROLS LINKAGE ASSEMBLY COMPONENT MAINTENANCE MANUAL 27-31-52

REVISION NO. 4 DATED JAN 01/89

HIGHLIGHTS

Pages which have been added or revised are outlined below together with the highlights of the revision. Remove and insert the affected pages as listed and enter Revision No. and date on the Record of Revision Sheet.

Remove and discard Repair 3-1, pages 601-602.

CHAPTER/SECTION AND PAGE NO.	DESCRIPTION OF CHANGE
TITLE PAGE 1	Changed title to correspond to production drawing nomenclature.
CONTENTS 1	Changed Special Tools listing to indicate none required.
INTRODUCTION 1	Deleted unnecessary verification statement.
301	Added missing item No. reference.
REPAIR-GEN 601 REPAIR 1-1 601-602	Combined Repair 3-1 with Repair 1-1 to cover almost identical shaft assemblies.
REPAIR-GEN 601	Added Chapter 20 Standard Practices references.
REPAIR-GEN 602	Updated True Position Dimensioning symbols.
REPAIR 2-1 601-603	Added bolthole repair instructions for lever assembly.

CHAPTER/SECTION AND PAGE NO.	DESCRIPTION OF CHANGE
701	Changed procedure to provide proper installation sequence.
801-802	Identified mating installation parts.

BOEING

COMPONENT
MAINTENANCE MANUAL

HORIZONTAL STABILIZER CENTER ELEVATOR CONTROLS LINKAGE ASSEMBLY

PART NUMBER 251T2600-9,-10

COMPONENT MAINTENANCE MANUAL
WITH
ILLUSTRATED PARTS LIST

80829

REVISION RECORD

- Retain this record in front of manual. On receipt of revision, insert revised pages in the manual, and enter revision number, date inserted and initial.

REVISION NUMBER	REVISION DATE	DATE FILED	BY

REVISION NUMBER	REVISION DATE	DATE FILED	BY

TEMPORARY REVISION AND SERVICE BULLETIN RECORD

BOEING SERVICE BULLETIN	BOEING TEMPORARY REVISION	OTHER DIRECTIVE	DATE OF INCORPORATION INTO MANUAL
		PRR B10716	OCT 10/83

PAGE	DATE	CODE
27-31-52		
TITLE PAGE		
*1	JAN 01/89	01.1
2	BLANK	
REVISION RECORD		
1	JUL 10/83	01
2	BLANK	
TR & SB RECORD		
1	OCT 10/83	01.1
2	BLANK	
LIST OF EFFECTIVE PAGES		
*1	JAN 01/89	01
THRU LAST PAGE		
CONTENTS		
*1	JAN 01/89	01.1
2	BLANK	
INTRODUCTION		
*1	JAN 01/89	01.101
2	BLANK	
DESCRIPTION & OPERATION		
1	OCT 10/83	01.1
2	BLANK	
DISASSEMBLY		
*301	JAN 01/89	01.1
302	BLANK	
CLEANING		
401	JUL 10/83	01
402	BLANK	
CHECK		
501	JUL 10/83	01
502	BLANK	

PAGE	DATE	CODE
REPAIR-GENERAL		
*601	JAN 01/89	01.1
*602	JAN 01/89	01.1
REPAIR 1-1		
*601	JAN 01/89	01.1
*602	JAN 01/89	01.1
REPAIR 2-1		
*601	JAN 01/89	01.1
*602	JAN 01/89	01.1
*603	JAN 01/89	01.1
*604	BLANK	
ASSEMBLY		
*701	JAN 01/89	01.1
702	BLANK	
FITS AND CLEARANCES		
*801	JAN 01/89	01.1
*802	JAN 01/89	01.1
803	JUL 10/85	01.101
804	BLANK	
ILLUSTRATED PARTS LIST		
1001	JUL 10/83	01
1002	JUL 10/83	01.1
1003	JUL 10/83	01.1
1004	JUL 10/83	01.1
1005	JUL 10/83	01.1
1006	JUL 10/83	01.1
1007	JUL 10/83	01.1
1008	BLANK	

* = REVISED, ADDED OR DELETED

TABLE OF CONTENTS

INTRODUCTION

The instructions in this manual provide the information necessary to perform maintenance functions ranging from simple checks and replacement to complete shop-type repair.

This manual is divided into separate sections:

1. Title Page
2. Record of Revisions
3. Temporary Revisions & Service Bulletin Record
4. List of Effective Pages
5. Table of Contents
6. Introduction
7. Procedures & IPL Sections

Refer to the Table of Contents for the page location of applicable sections. An asterisked flagnote *[ ] in place of the page number indicates that no special instructions are provided since the function can be performed using standard industry practices.

The beginning of the REPAIR section includes a list of the separate repairs, a list of applicable standard Boeing practices, and an explanation of the True Position Dimensioning symbols used.

An explanation of the use of the Illustrated Parts List is provided in the Introduction to that section.

All weights and measurements used in the manual are in English units, unless otherwise stated. When metric equivalents are given they will be in parentheses following the English units.

Design changes, optional parts, configuration differences and Service Bulletin modifications create alternate part numbers. These are identified in the Illustrated Parts List (IPL) by adding an alphabetical character to the basic item number. The resulting item number is called an alpha-variant. Throughout the manual, IPL basic item number references also apply to alpha-variants unless otherwise indicated.

HORIZONTAL STABILIZER CENTER ELEVATOR CONTROLS LINKAGE ASSEMBLY

DESCRIPTION AND OPERATION

1. The linkage assembly consists of an input lever assembly attached to a shaft assembly. The lever assembly receives its input from the aft quadrant assembly and transmits this motion thru linkage to the elevator power control actuator.

2. Leading Particulars (Approximate)

   Length -- 9 inches
   Width -- 11 inches
   Height -- 7 inches
   Weight -- 4 pounds

DISASSEMBLY

NOTE: Disassemble this component only as necessary to complete fault isolation, determine the serviceability of parts, perform required repairs, and restore the unit to serviceable condition.

1. Disassembly (IPL Fig. 1)

   A. Remove bolt (10), bearing (15), spacer (25), bushing (20) and nut (30).

      NOTE: Do not remove rivets (65), plate (70), stop pads (75), bushings (80, 85) from shaft (90 or 90A) unless necessary for repair or replacement. Do not disassemble lever assy (35) unless necessary for repair or replacement.

CLEANING

1. Clean all parts, except bearings, using standard industry practices per 20-30-03.

2. Clean teflon sealed bearings (15, 45) according to manufacturer's instruction.

CHECK

1. Check all parts for obvious defects in accordance with standard industry practices.

2. Refer to FITS AND CLEARANCES for design dimensions and wear limits.

3. Penetrant check the following parts (Ref IPL Fig. 1) per 20-20-02.

   A. Lever (50, 55)

   B. Shaft (90, 90A)

REPAIR - GENERAL

1. Content

A. Repair, refinish and replacement procedures are included in separate repair sections as follows:

P/N	NAME	REPAIR
251T2610 251T2618	SHAFT	1-1
251T2612	LEVER	2-1
--	(DELETED)	

2. Standard Practices

A. Refer to the following standard practices as applicable, for details of procedures in individual repairs.

20-10-05 Application and Finishing of Plasma Flame Sprayed Coatings
20-30-02 Stripping of Protective Finishes
20-30-03 General Cleaning Procedures
20-41-01 Decoding Table for Boeing Finish Codes
20-41-02 Application of Chemical and Solvent Resistant Finishes
20-42-03 Hard Chrome Plating
20-42-05 Bright Cadmium Plating
20-43-01 Chromic Acid Anodizing
20-50-03 Bearing Installation and Retention

3. Materials

NOTE: Equivalent substitutes may be used.

A. Grease -- MIL-G-23827 (Ref 20-60-03)

B. Primer -- BMS 10-11, type 1 (Ref 20-60-02)

C. Sealant -- BMS 5-95 (Ref 20-60-04)

4. Dimensioning Symbols

A. Standard True Position Dimensioning Symbols used in applicable repair procedures are shown in Fig. 601.

—	STRAIGHTNESS
▱	FLATNESS
⊥	PERPENDICULARITY (OR SQUARENESS)
//	PARALLELISM
○	ROUNDNESS
⌭	CYLINDRICITY
⌒	PROFILE OF A LINE
⌓	PROFILE OF A SURFACE
◎	CONCENTRICITY
⌯	SYMMETRY
∠	ANGULARITY
↗	RUNOUT
⌰	TOTAL RUNOUT
⌴	COUNTERBORE OR SPOTFACE
⌵	COUNTERSINK
⌖	THEORETICAL EXACT POSITION OF A FEATURE (TRUE POSITION)
∅	DIAMETER
S ∅	SPHERICAL DIAMETER
R	RADIUS
SR	SPHERICAL RADIUS
()	REFERENCE
BASIC (BSC) OR [DIM]	A THEORETICALLY EXACT DIMENSION USED TO DESCRIBE SIZE, SHAPE OR LOCATION OF A FEATURE FROM WHICH PERMISSIBLE VARIATIONS ARE ESTABLISHED BY TOLERANCES ON OTHER DIMENSIONS OR NOTES.
[-A-]	DATUM
Ⓜ	MAXIMUM MATERIAL CONDITION (MMC)
Ⓛ	LEAST MATERIAL CONDITION (LMC)
Ⓢ	REGARDLESS OF FEATURE SIZE (RFS)
Ⓟ	PROJECTED TOLERANCE ZONE
FIM	FULL INDICATOR MOVEMENT

EXAMPLES

— | 0.002	STRAIGHT WITHIN 0.002
⊥ | B | 0.002	PERPENDICULAR TO B WITHIN 0.002
// | A | 0.002	PARALLEL TO A WITHIN 0.002
○ | 0.002	ROUND WITHIN 0.002
⌭ | 0.010	CYLINDRICAL SURFACE MUST LIE BETWEEN TWO CONCENTRIC CYLINDERS, ONE OF WHICH HAS A RADIUS 0.010 INCH GREATER THAN THE OTHER
⌒ | A | 0.006	EACH LINE ELEMENT OF THE SURFACE AT ANY CROSS SECTION MUST LIE BETWEEN TWO PROFILE BOUNDARIES 0.006 INCH APART RELATIVE TO DATUM PLANE A
⌓ | A | 0.020	SURFACES MUST LIE WITHIN PARALLEL BOUNDARIES 0.02 INCH APART AND EQUALLY DISPOSED ABOUT TRUE PROFILE
◎ | C | ∅ 0.0005	CONCENTRIC TO C WITHIN 0.0005 DIAMETER
⌯ | A | 0.010	SYMMETRICAL WITH A WITHIN 0.010
∠ | A | 0.005	ANGULAR TOLERANCE 0.005 WITH A
⌖ | B | ∅ 0.002 Ⓢ	LOCATED AT TRUE POSITION WITHIN 0.002 DIA RELATIVE TO DATUM B, REGARDLESS OF FEATURE SIZE
⊥ | A | ∅ 0.010 Ⓜ 0.510 Ⓟ	AXIS IS TOTALLY WITHIN A CYLINDER OF 0.010-INCH DIAMETER, PERPENDICULAR TO, AND EXTENDING 0.510-INCH ABOVE, DATUM A, MAXIMUM MATERIAL CONDITION
[2.000] OR 2.000 BSC	EXACT DIMENSION IS 2.000

(NOTE THAT ⌓ | A | 0.020 MAY ALSO APPEAR AS ⌓ | 0.020 | A)

True Position Dimensioning Symbols
Figure 601

SHAFT ASSEMBLY - REPAIR 1-1

251T2610-1
251T2618-1

NOTE: Refer to REPAIR-GEN for list of applicable standard practices. For repair of surfaces which may only require restoration of original finish, refer to Refinish instructions, Fig. 601.

1. Bushing Replacement (Fig. 601)

   A. Remove bushings (80, 85).

   B. Install bushings (85) per 20-50-03 except use wet sealant.

   C. Machine bushing bores as shown.

   D. Fillet seal bushing flanges with sealant.

   E. Install bushing (80) and stake with tool ST922C-4 per 20-50-03 except use wet sealant.

   F. Fillet seal bushing flanges with sealant.

2. Stop Pad Replacement

   A. Remove stop pad (75).

   B. Install new stop pad (75) with sealant. Press until flange is seated.

   C. Fillet seal stop pad flange with sealant.

3. Plate Replacement (251T2618-1 only)

   A. Remove rivets (65).

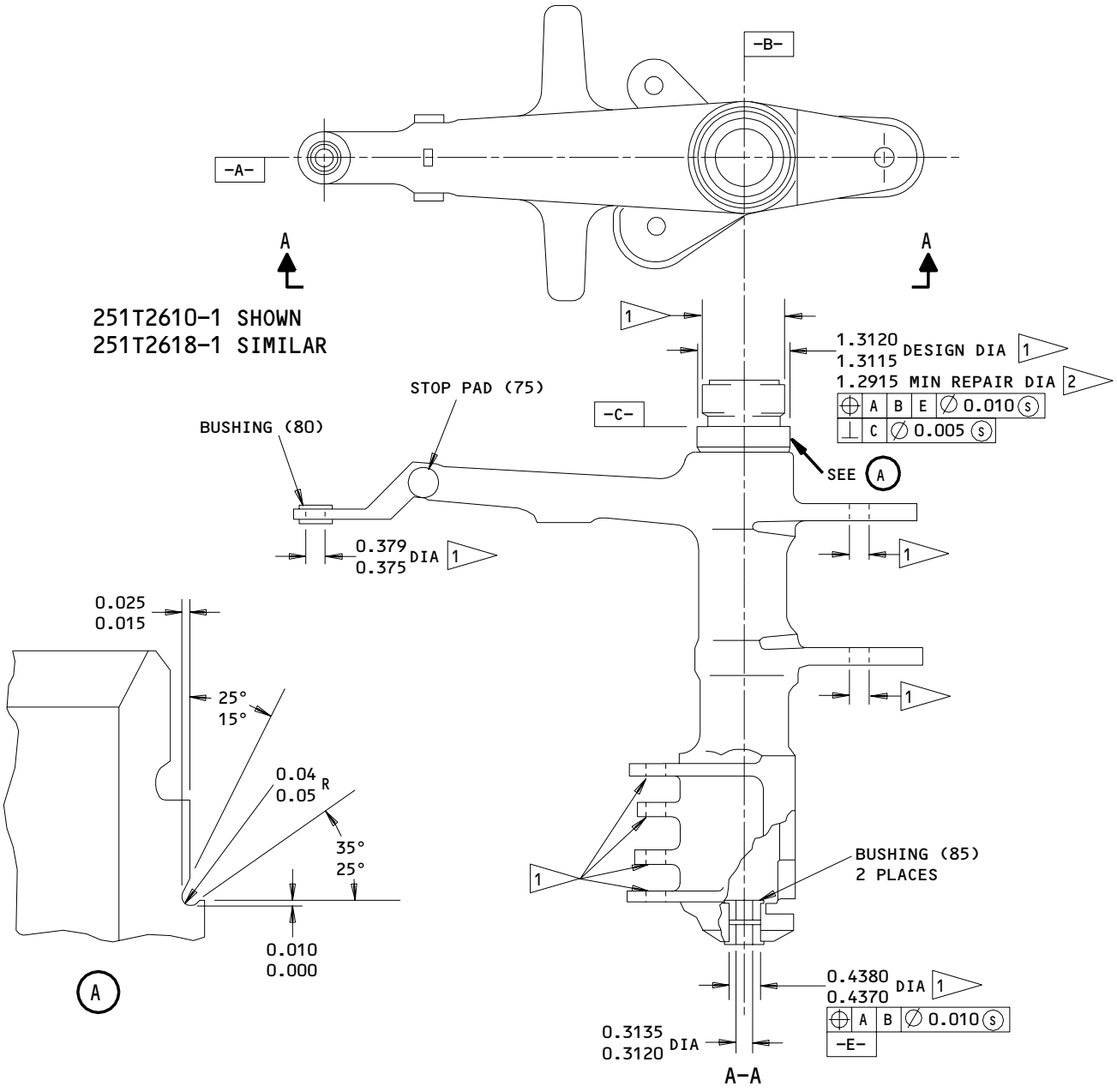
   B. Remove plate (70) from shaft (90A).

   C. Install plate (70) on shaft (90A) and secure with rivets (65).

4. Bearing Seat Repair (Fig. 601)

   A. Machine bearing seat as required, within repair limit shown, to remove defects.

   B. Build up repaired surface with chrome plate and grind to design dimension and finish shown. Observe 0.010 inch maximum plating thickness. Optional: build up with plasma flame spray BMS 10-67, type 10 and machine to design dimension and finish shown.



REFINISH

SHAFT (90) -- CHROMIC ACID ANODIZE (F-17.04) AND APPLY TWO COATS OF PRIMER BMS 10-11, TYPE 1 (F-20.03) ALL OVER EXCEPT AS NOTED

1 NO PRIMER ON THIS SURFACE

2 REPAIR DIA FOR BUILD UP WITH CHROME PLATE (F-15.03) OR PLASMA FLAME SPRAY, BMS 10-67, TYPE 10

REPAIR

REF 2

125 MACHINED SURFACES

BREAK SHARP EDGES 0.008 R

MATERIAL: AL ALLOY

ALL DIMENSIONS ARE IN INCHES

ITEM NUMBERS REFER TO IPL FIG. 1

251T2610-1
251T2618-1
Shaft Repair
Figure 601

80836

LEVER ASSEMBLY - REPAIR 2-1

251T2612-11 thru -14

NOTE: Refer to REPAIR-GEN for list of applicable standard practices. For repair of surfaces which may only require restoration of original finish, refer to Refinish instructions, Fig. 601.

1. Bearing Replacement (Fig. 601)

   A. Remove bearing (45).

   B. Install new bearing and roller swage per 20-50-03.

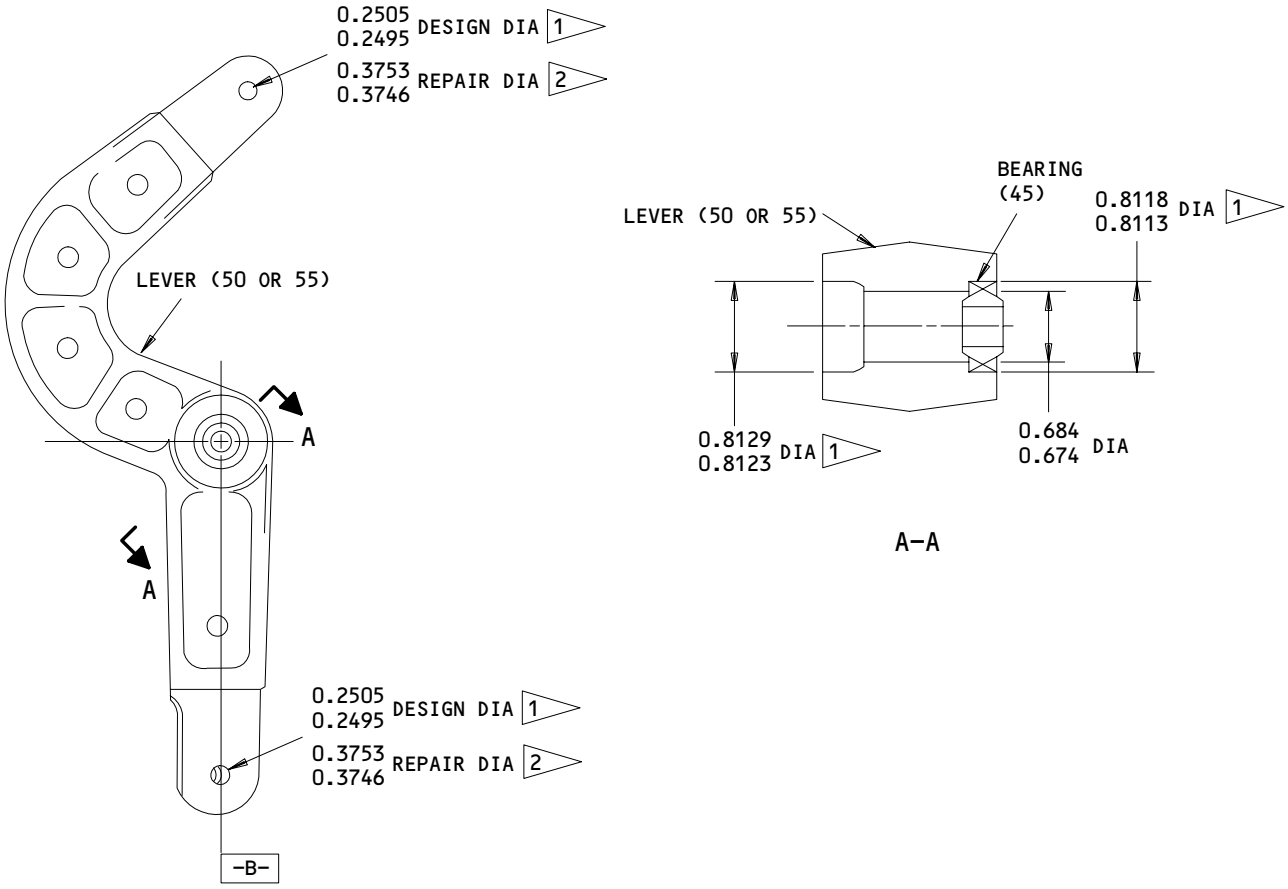
2. Bolthole Repair (Fig. 601)

   A. Machine boltholes, in sets of two, to repair diameter shown.

   B. Make repair bushings per Fig. 602 or use NAS537-4P016 or equivalent cadmium-plated steel bushing.

   C. Install bushings per 20-50-03. Bushings shall be flush on inside surface of fork and shall protrude 0.000-0.015 inch on outside.

   D. Machine bushings to design dimension shown.



REFINISH

LEVER (50,55) -- CHROMIC ACID ANODIZE (F-17.04), PLUS APPLY TWO COATS OF PRIMER BMS 10-11, TYPE 1, (F-20.03) ALL OVER EXCEPT AS NOTED

1 NO PRIMER ON THIS SURFACE

2 REPAIR LIMIT FOR INSTALLATION OF REPAIR BUSHING

REPAIR

REF 2

MATERIAL: AL ALLOY

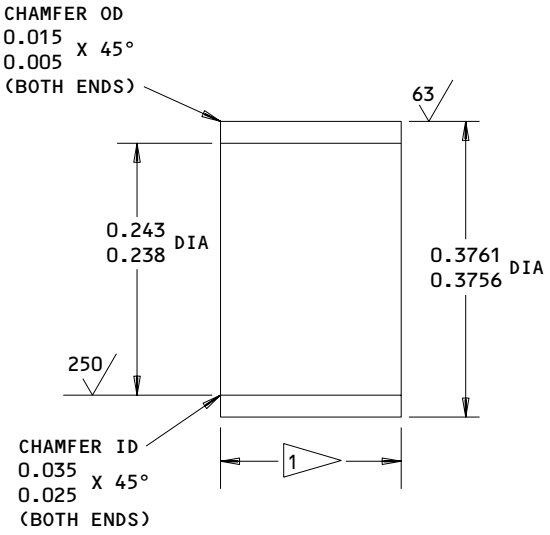
ALL DIMENSIONS ARE IN INCHES

ITEM NUMBERS REFER TO IPL FIG. 1

251T2612-11 THRU -14

Lever Assembly Repair
Figure 601

80838



MANUFACTURED BUSHING

1 BUSHING LENGTH EQUAL TO LUG THICKNESS PLUS 0.000–0.015

MATERIAL: 4130 STEEL, 125–145 KSI
FINISH: CADMIUM PLATE (F–15.06)
ALL DIMENSIONS ARE IN INCHES

251T2612–11 THRU –14
Repair Bushing for Lever Assembly
Figure 602

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ASSEMBLY

1. Material

   NOTE: Equivalent substitutes may be used.

   A. Corrosion Preventive Compound -- MIL-C-11796, Class 3 (Ref 20-60-03)

2. Assembly (IPL Fig. 1)

   A. Install spacer (25), bearing (15) into input lever assembly (35) per 20-50-03. Install bearing (15) with corrosion preventive compound.

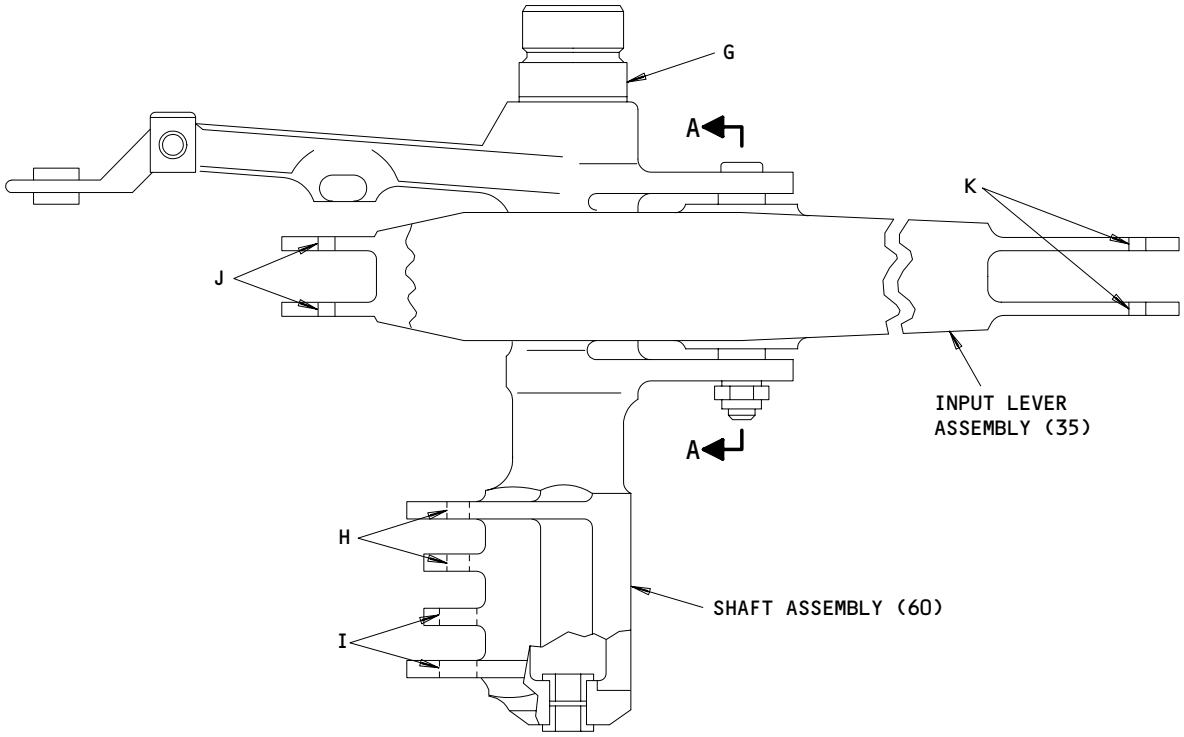
   B. Apply corrosion preventive compound to all surfaces of bolt (10), bushing (20) and nut (30).

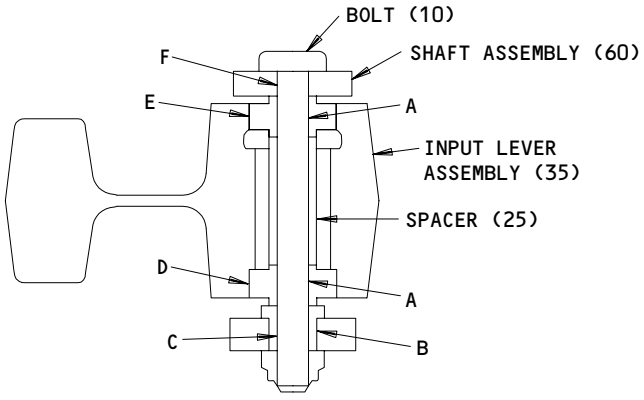
   C. Install input lever assembly (35) on shaft assembly (60) with bushing (20), bolt (10) and nut (30). Tighten nut (30) to 50-60 pound-inches.

3. Storage

   A. Prepare and store component in accordance with standard industry practices.

FITS AND CLEARANCES





A-A

Fits and Clearances
Figure 801 (Sheet 1)

Ref Letter Fig.801	Mating Item No. IPL Fig.	Design Dimension				Service Wear Limit		
		Dimension		Assembly Clearance 1		Dimension		Maximum Clearance
		Min	Max	Min	Max	Min	Max	
A	ID 15,45	0.3120	0.3125	0.0000	0.0015			
	OD 10	0.3110	0.3120					
B	ID 90	0.4388	0.4393	0.0002	0.0012		0.4398	0.0020
	OD 20	0.4381	0.4386			0.4376		
C	ID 20	0.3125	0.3140	0.0005	0.0030		0.3145	0.0040
	OD 10	0.3110	0.3120			0.3105		
D	ID 50	0.8113	0.8118	-0.0012	-0.0002		0.8123	-0.0002
	OD 45	0.8120	0.8125			0.8115		
E	ID 35	0.8123	0.8129	-0.0002	0.0009		0.8134	0.0020
	OD 15	0.8120	0.8125			0.8115		
F	ID 60	0.3120	0.3130	0.0000	0.0020		0.3135	0.0030
	OD 10	0.3110	0.3120			0.3105		
G	ID 2	1.3120	1.3125	0.0000	0.0010		1.3130	0.0020
	OD 90	1.3115	1.3120			1.3110		
H	ID 90	0.2495	0.2505	0.0000	0.0020		0.2510	0.0030
	OD 3	0.2485	0.2495			0.2480		
I	ID 90	0.3763	0.3768	0.0002	0.0012		0.3773	0.0020
	OD 4	0.3756	0.3761			0.3751		
J	ID 35	0.2495	0.2505	0.0000	0.0020		0.2510	0.0030
	OD 3	0.2485	0.2495			0.2480		
K	ID 35	0.2495	0.2505	0.0000	0.0020		0.2510	0.0030
	OD 3	0.2485	0.2495			0.2480		

1 NEGATIVE VALUES DENOTE INTERFERENCE FIT
2 INSTALLATION PART BACB10AW21
3 INSTALLATION PART BACB30NF4-( )
4 INSTALLATION PART NAS77-4-25
ALL DIMENSIONS ARE IN INCHES

Fits and Clearances
Figure 801 (Sheet 2)

80841

FOR TORQUE VALUES OF STANDARD FASTENERS. REFER TO 20-50-01			
ITEM NO. IPL Fig. 1	NAME	TORQUE	
		POUND-INCHES	POUND-FEET
30	Nut	50-60	

Torque Table
Figure 802

ILLUSTRATED PARTS LIST

1. This section lists and illustrates replaceable or repairable component parts. The Illustrated Parts Catalog contains a complete explanation of the Boeing part numbering system.

2. Indentures show parts relationships as follows:

    Assembly
      Detail Parts for Assembly
      Subassembly
      Attaching Parts for Subassembly
        Detail Parts for Subassembly

    Detail Installation Parts (Included only if installation parts may be returned to shop as part of assembly)

3. One use code letter (A, B, C, etc.) is assigned in the EFF CODE column for each variation of top assembly. All listed parts are used on all top assemblies except when limitations are shown by use code letter opposite individual part entries.

4. Letter suffixes (alpha-variants) are added to item numbers for optional parts, Service Bulletin modification parts, configuration differences (except left- and right-hand parts), product improvement parts, and parts added between two sequential item numbers. The alpha-variant is not shown on illustrations when appearance and location of all variants of the part are the same.

5. Service Bulletin modifications are shown by the notations PRE SB XXXX and POST SB XXXX.

    A. When a new top assembly part number is assigned by Service Bulletin, the notations appear at the top assembly level only. The configuration differences at detail part level are then shown by use code letter.

    B. When the top assembly part number is not changed by the Service Bulletin, the notations appear at the detail part level.

6. Parts Interchangeability

Optional (OPT)	The parts are optional to and interchangeable with other parts having the same item number.
Supersedes, Superseded By (SUPSDS, SUPSD BY)	The part supersedes and is not interchangeable with the original part.
Replaces, Replaced By (REPLS, REPLD BY)	The part replaces and is interchangeable with, or is an alternate to, the original part.

VENDORS

06710 VALLEY-TODECO INCORPORATED
12975 BRADLEY AVENUE
SYLMAR,CALIFORNIA 91342

06725 AIR INDUSTRIES CORPORATION
12570 KNOTT STREET
GARDEN GROVE, CALIFORNIA 92641

06950 VSI CORP SCREWCORP DIV
13001 EAST TEMPLE AVENUE
CITY OF INDUSTRY, CALIFORNIA 91746

08524 DEUTSCH FASTENER CORPORATION
PO BOX 92925 7001 WEST IMPERIAL HIGHWAY
LOS ANGELES, CALIFORNIA 90045

17943 FEDERAL MANUFACTURING CORPORATION
6910 FARMDALE AVENUE
NORTH HOLLYWOOD, CALIFORNIA 91605

21335 TEXTRON INC FAFNIR BEARING DIVISION
37 BOOTH STREET
NEW BRITAIN, CONNECTICUT 06050

23294 AVALON MACHINE PRODUCTS INC
15337 ALLEN STREET
PARAMOUNT, CALIFORNIA 90723

27624 PAUL R BRILES INC P.B. FASTENER DIV
1700 WEST 132ND STREET PO BOX 1157
GARDENA, CALIFORNIA 90249

30163 DAYRON CORP
333 MAGUIRE BLVD PO BOX 20394
ORLANDO, FLORIDA 32814

38443 TRW INC BEARING DIV
402 CHANDLER STREET
JAMESTOWN, NEW YORK 14701

43991 FAG BEARING INCORPORATED
HAMILTON AVENUE
STAMFORD, CONNECTICUT 06904

VENDORS

52828 REPUBLIC FASTENER MFG CORP
1300 RANCHO CONEJO BLVD
NEWBURY PARK, CALIFORNIA 91320

56878 SPS TECHNOLOGIES INC
HIGHLAND AVENUE
JENKINTOWN, PENNSYLVANIA 19046

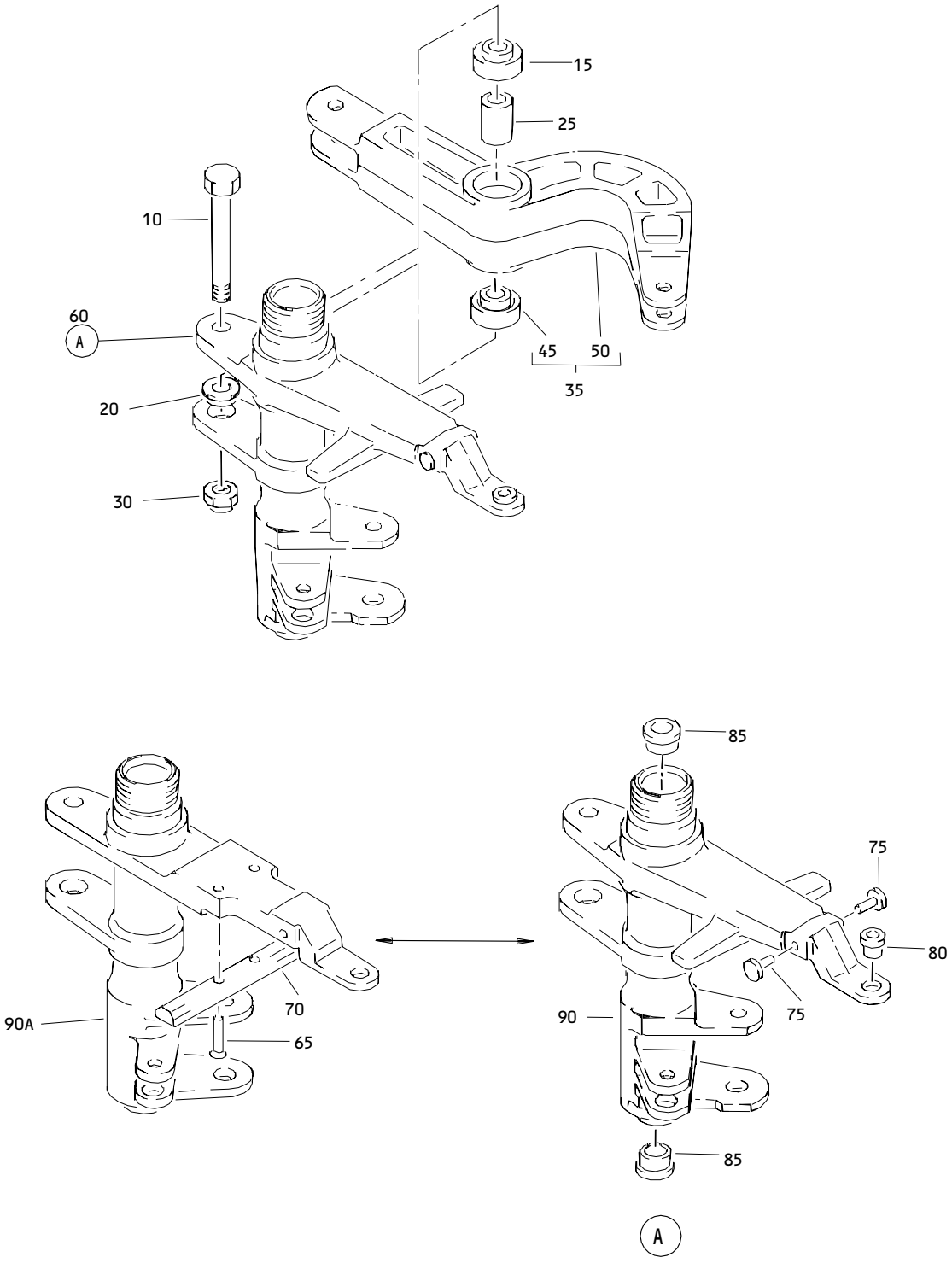
70265 ALL POWER MANUFACTURING COMPANY
13141 MOLETTE STREET
SANTE FE SPRINGS, CALIFORNIA 90670

80539 SPS TECHNOLOGIES INC AEROSPACE PRODUCTS DIV
2701 SOUTH HARBOR BOULEVARD PO BOX 1259
SANTA ANA, CALIFORNIA 92702

92215 VOI-SHAN DIV OF VSI CORP
8463 HIGUERA STREET
CULVER CITY, CALIFORNIA 90230

94892 MASTER MACHINE PRODUCTS CORPORATION
2069 RANDOLPH STREET
HUNTINGTON PARK, CALIFORNIA 90255

97928 LITTON FASTENING SYSTEMS DIV OF LITTON SYSTEMS INC
3969 PARAMONT BOULEVARD
LAKEWOOD, CALIFORNIA 90712



Horizontal Stabilizer Center Elevator Controls Linkage Assembly
Figure 1

80726

FIG. & ITEM	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE 1234567	EFF CODE	QTY PER ASSY
01-					
-1	251T2600-9		LINKAGE ASSY-HORIZ STAB. CTR ELEVATOR CONT (LH)	A	RF
-5	251T2600-10		LINKAGE ASSY-HORIZ STAB. CTR ELEVATOR CONT (RH)	B	RF
10	BACB30NF5-37		.BOLT- (V06710) (SPEC BACB30NF5-37) (OPT BACB30NF5-37 (V06725)) (OPT BACB30NF5-37 (V06950)) (OPT BACB30NF5-37 (V08524)) (OPT BACB30NF5-37 (V17943)) (OPT BACB30NF5-37 (V27624)) (OPT BACB30NF5-37 (V56878)) (OPT BACB30NF5-37 (V80539)) (OPT BACB30NF5-37 (V92215)) (OPT BACB30NF5-37 (V97928))		1
15	KP5A		.BEARING- (V38443) (SPEC BACB10BX5) (OPT KP5AFS428 (V21335)) (OPT KP5A2TS (V43991)) (OPT LLKP5A (V38443)) (OPT KP5AG27 (V30163))		1

FIG. & ITEM	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE 1234567	EFF CODE	QTY PER ASSY
01-					
20	NAS77-5-32		.BUSHING		1
25	NAS43DD5-72		.SPACER		1
30	BRH10-5		.NUT- (V52828) (SPEC BACN10JC5)		1
35	251T2612-11		.LEVER ASSY-PILOT INPUT (OPT ITEM 35A)	A	1
-35A	251T2612-13		.LEVER ASSY-PILOT INPUT (OPT ITEM 35)	A	1
-40	251T2612-12		.LEVER ASSY-PILOT INPUT (OPT ITEM 40A)	B	1
-40A	251T2612-14		.LEVER ASSY-PILOT INPUT (OPT ITEM 40)	B	1
45	KP5A		..BEARING- (V38443) (SPEC BACB10BX5) (SEE ITEM 15 FOR OPT PARTS)		1
50	251T2612-15		..LEVER- (USED ON ITEM 35)	A	1
-50A	251T2612-17		..LEVER- (USED ON ITEM 35A)	A	1
-55	251T2612-16		..LEVER- (USED ON ITEM 40)	B	1
-55A	251T2612-18		..LEVER- (USED ON ITEM 40A)	B	1
60	251T2610-1		.SHAFT ASSY- (OPT ITEM 60A)		1
-60A	251T2618-1		.SHAFT ASSY- (OPT ITEM 60)		1
65	BACR15BA5KE		..RIVET- (USED ON ITEM 60A)		2
70	251T2618-3		..PLATE- (USED ON ITEM 60A)		1
75	251T2626-1		..PAD-STOP		2

FIG. & ITEM	PART NO.	AIRLINE PART NUMBER	NOMENCLATURE 1234567	EFF CODE	QTY PER ASSY
01-					
80	BACB28B4-225P		..BUSHING- (V23294) (SPEC BACB28B4-225P) (OPT BACB28B4-225P (V70265)) (OPT BACB28B4-225P (V94892))		1
85	BACB28W5C022		..BUSHING- (V23294) (SPEC BACB28W5C022) (OPT BACB28W5C022 (V70265)) (OPT BACB28W5C022 (V94892))		2
90	251T2610-2		..SHAFT- (USED ON ITEM 60)		1
90A	251T2618-2		..SHAFT- (USED ON ITEM 60A)		1